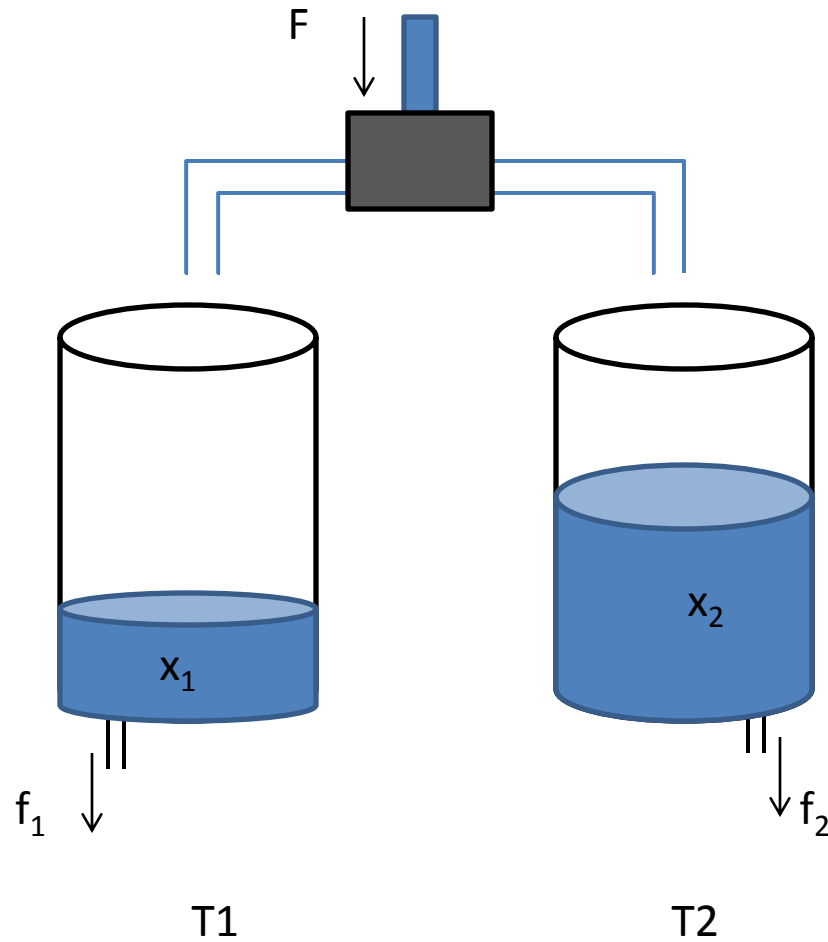


# Embedded and Cyber-Physical Systems

- Modeling of continuous dynamics-

Reference book: Edward A. Lee and Pravin Varaiya, [Structure and Interpretation of Signals and Systems](#), Second Edition, LeeVaraiya.org, ISBN 978-0-578-07719-2, 2011.

# A two-tank example

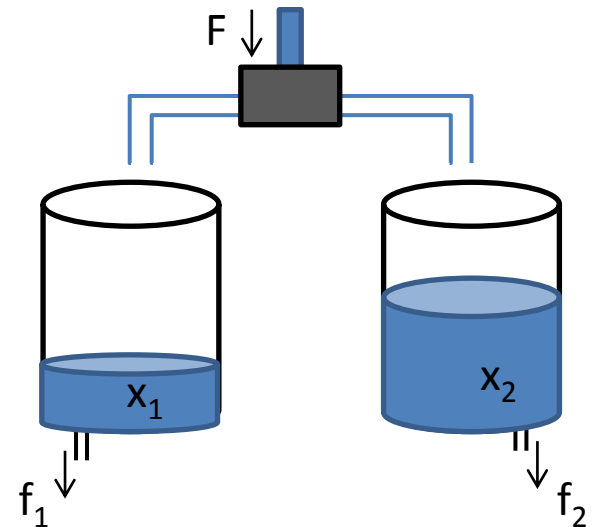


# The problem

- Setting:
  - $F(t) = c$  (constant)
  - $f_1(0) = f_2(0) = 0$
  - $x_1(0) > 0, x_2(0) > 0$
  - $f_1(t) = c_1, f_2(t) = c_2$  for  $t > 0$
  - $c_1 < c$  and  $c_2 < c$
  - $F$  can be switched either to  $T1$  or to  $T2$
- Requirement: find a suitable switching to keep both levels above zero.

# The Cyber Part

- Possible solution: a discrete controller
  - if (when)  $x_i == 0$ , switch to  $T_i$  ( $i=1,2$ )
- Advantage: Minimally interventive
- Is this a viable controller?



# The Physical Part

- Principle of mass conservation (assume isolated system, no work)

**Input mass = stored mass + output mass**

$$F \cdot t = M_1(t) + M_2(t) + (f_1 + f_2) \cdot t$$

$$F = \frac{dm}{dt}$$

$$M_1(t) + M_2(t) = (F - f_1 - f_2) \cdot t$$

What if  $(F - f_1 - f_2) < 0$  ?

# A mathematical model

$$(1) \begin{cases} \dot{x}_1 = F - f_1 \\ \dot{x}_2 = -f_2 \end{cases}$$

$$(2) \begin{cases} \dot{x}_1 = -f_1 \\ \dot{x}_2 = F - f_2 \end{cases}$$

# Solution

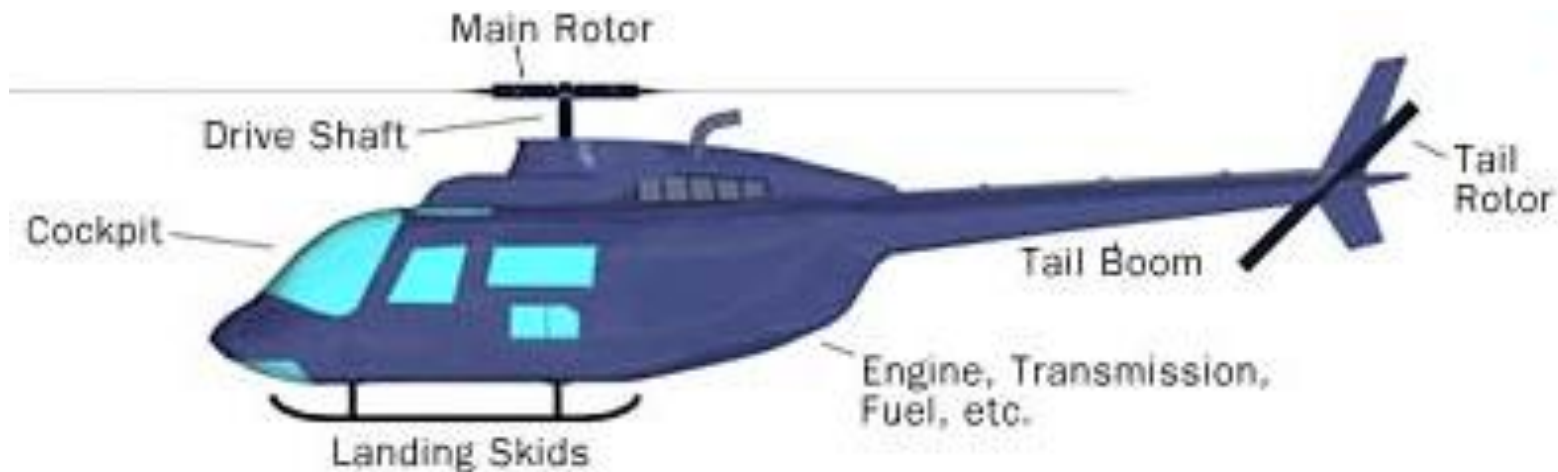
$$\begin{cases} x_1(t) = x_1^0 + (F - f_1) \cdot \Delta t_1 + (-f_1) \cdot \Delta t_2 \\ x_2(t) = x_2^0 + (-f_2) \cdot \Delta t_1 + (F - f_2) \cdot \Delta t_2 \end{cases}$$

Does a viable controller exist?

Check this:

$$\begin{cases} (F - f_1) \cdot \Delta t_1 + (-f_1) \cdot \Delta t_2 \geq 0 \\ (-f_2) \cdot \Delta t_1 + (F - f_2) \cdot \Delta t_2 \geq 0 \end{cases}$$

# Another Example: Modeling Helicopter Dynamics



**The Fundamental Parts of any Helicopter**

©2000 HowStuffWorks



# Feedback Control Problem

A helicopter without a tail rotor, like the one below, will spin uncontrollably. Why?

Control system problem:  
Apply torque using the tail rotor.



# Physics teaser: helicopter spin

- Watch the two videos, with helicopters that are similar in construction.

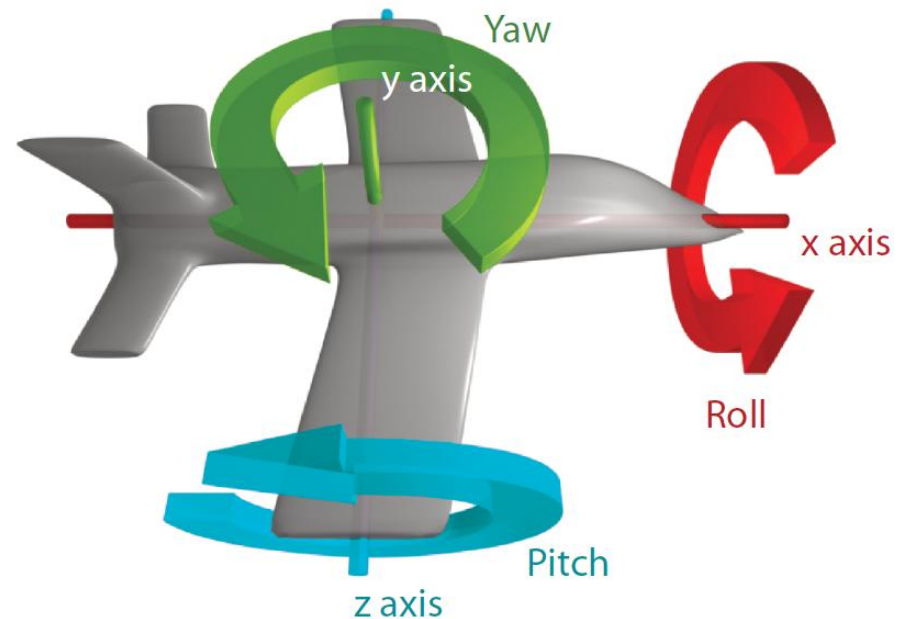
<http://www.youtube.com/watch?v=MhEXXgiIVuY>

[http://www.youtube.com/watch?v=ijErOpjx\\_Ws](http://www.youtube.com/watch?v=ijErOpjx_Ws)

- Why do they behave differently after loosing the tail rotor control?

# Modeling Physical Motion

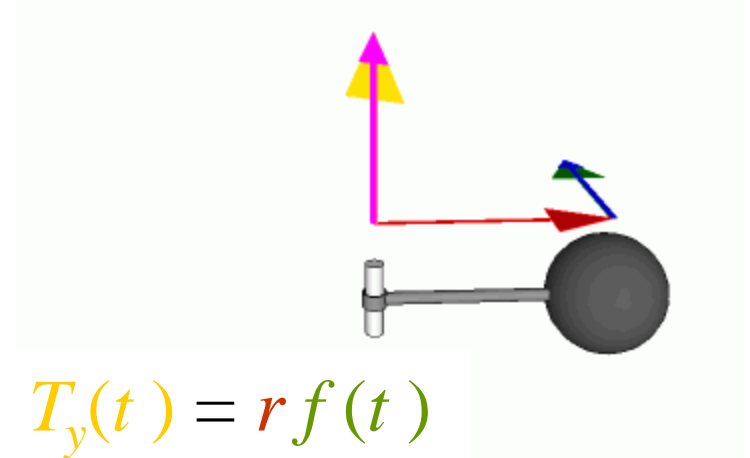
- Six degrees of freedom:
  - Position:  $x, y, z$
  - Orientation: pitch, yaw, roll



# Torque

For a point mass rotating around a fixed axis:

- radius of the arm:  $r \in \mathbb{R}$
- force orthogonal to arm:  $f \in \mathbb{R}$



- Just as force is a push or a pull, a torque is a twist.
- Units: Newton-meters/radian, Joules/radian

# Rotational Version of Newton's Second Law

$$\mathbf{T}(t) = \frac{d}{dt} \left( I(t) \dot{\theta}(t) \right),$$

where  $I(t)$  is a  $3 \times 3$  matrix called the moment of inertia tensor.

$$\begin{bmatrix} T_x(t) \\ T_y(t) \\ T_z(t) \end{bmatrix} = \frac{d}{dt} \left( \begin{bmatrix} I_{xx}(t) & I_{xy}(t) & I_{xz}(t) \\ I_{yx}(t) & I_{yy}(t) & I_{yz}(t) \\ I_{zx}(t) & I_{zy}(t) & I_{zz}(t) \end{bmatrix} \begin{bmatrix} \dot{\theta}_x(t) \\ \dot{\theta}_y(t) \\ \dot{\theta}_z(t) \end{bmatrix} \right)$$

Here, for example,  $T_y(t)$  is the net torque around the  $y$  axis (which would cause changes in yaw),  $I_{yx}(t)$  is the inertia that determines how acceleration around the  $x$  axis is related to torque around the  $y$  axis.

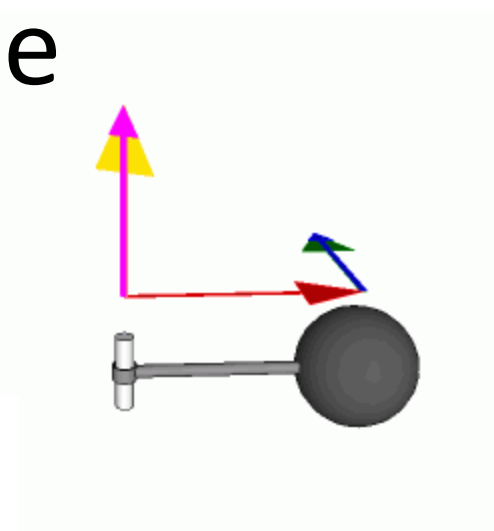
# Simple Example

Yaw dynamics:

$$T_y(t) = I_{yy}\ddot{\theta}_y(t)$$

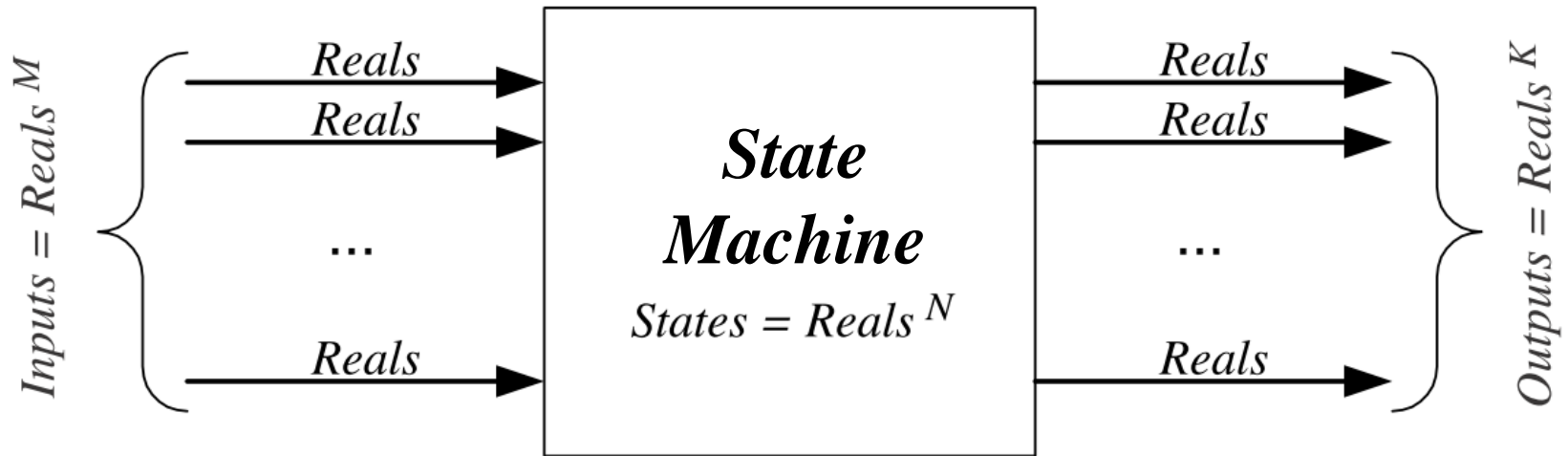
To account for initial angular velocity, write as

$$\dot{\theta}_y(t) = \dot{\theta}_y(0) + \frac{1}{I_{yy}} \int_0^t T_y(\tau) d\tau.$$



# Models

# Discrete model of systems



$StateMachine = (States, Inputs, Outputs, update, initialState)$

$s(n) \quad x(n) \quad y(n) \quad s(0)$

$s(0) = initialState,$

$\forall n \geq 0, (s(n+1), y(n)) = update(s(n), x(n))$

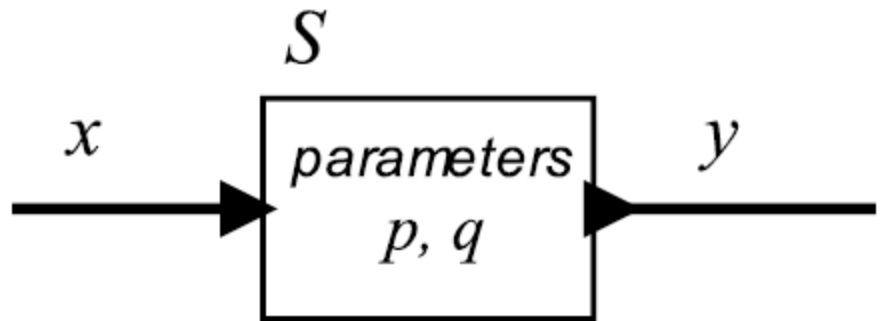


# Actor Model of Systems

- A *system* is a function that accepts an input *signal* and yields an output signal.

- The domain and range of the system function are sets of signals, which themselves are functions.

- Parameters may affect the definition of the function  $S$ .



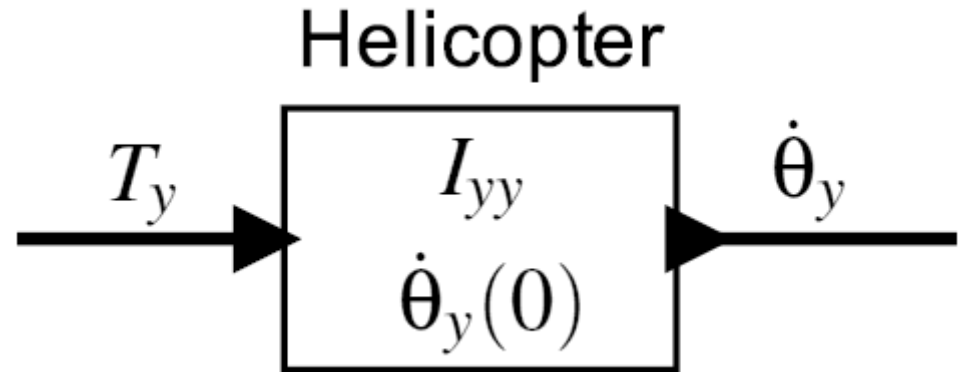
$$x: \mathbb{R} \rightarrow \mathbb{R}, \quad y: \mathbb{R} \rightarrow \mathbb{R}$$

$$S: X \rightarrow Y$$

$$X = Y = (\mathbb{R} \rightarrow \mathbb{R})$$

# Actor model of the helicopter

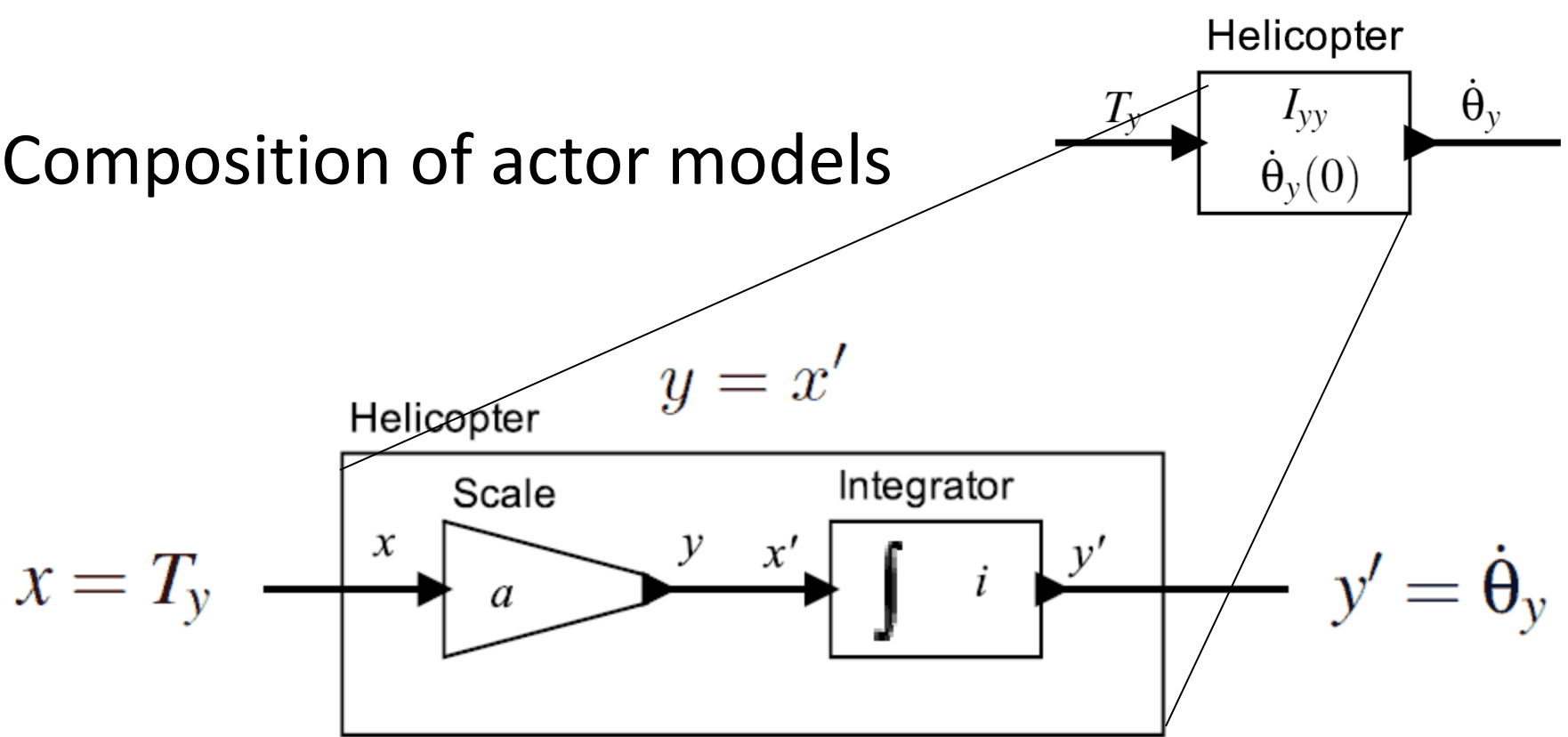
- Input is the net torque of the tail rotor. Output is the angular velocity around the  $y$  axis.



Parameters of the model are shown in the box. The input and output relation is given by the equation to the right.

$$\dot{\theta}_y(t) = \dot{\theta}_y(0) + \frac{1}{I_{yy}} \int_0^t T_y(\tau) d\tau$$

# Composition of actor models

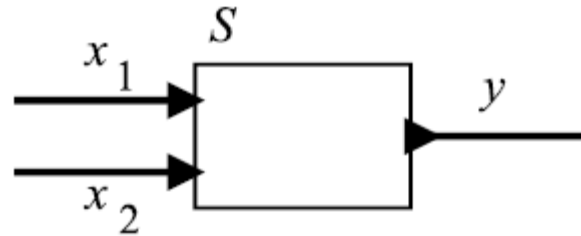


$$\forall t \in \mathbb{R}, \quad y(t) = ax(t) \quad y'(t) = i + \int_0^t x'(\tau) d\tau$$

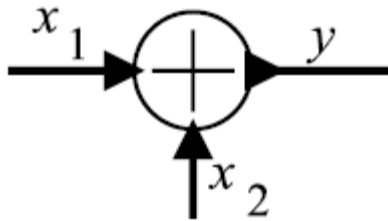
$$y = ax$$

$$a = 1/I_{yy} \quad i = \dot{\theta}_y(0)$$

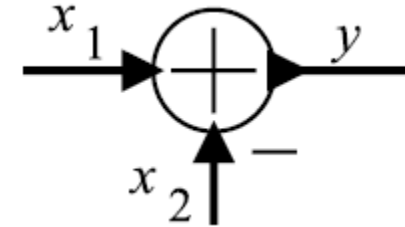
# Actor models with multiple inputs



$$S: (\mathbb{R} \rightarrow \mathbb{R})^2 \rightarrow (\mathbb{R} \rightarrow \mathbb{R})$$

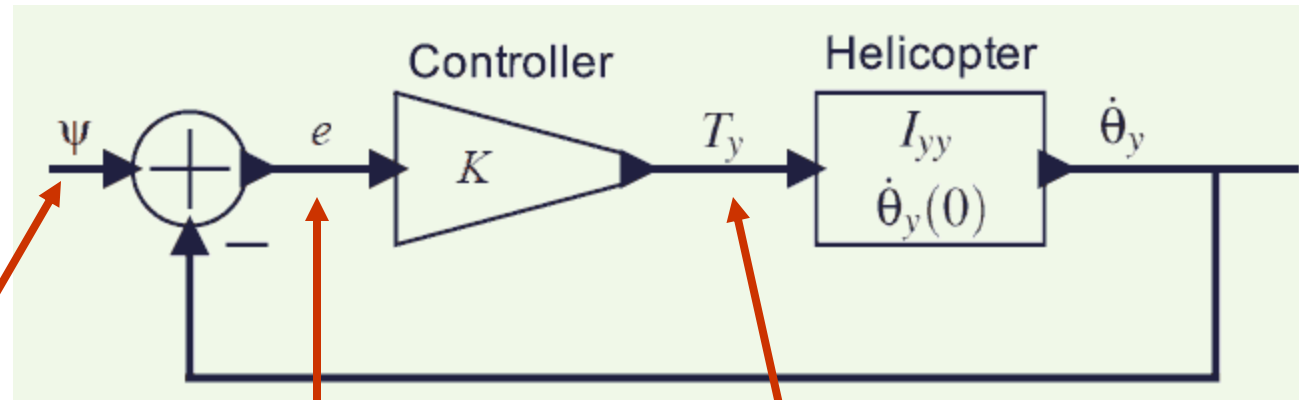


$$\forall t \in \mathbb{R}, \quad y(t) = x_1(t) + x_2(t)$$



$$(S(x_1, x_2))(t) = y(t) = x_1(t) - x_2(t)$$

# Proportional controller



desired  
angular  
velocity

error  
signal

net  
torque

$$e(t) = \psi(t) - \dot{\theta}_y(t)$$

$$T_y(t) = Ke(t)$$

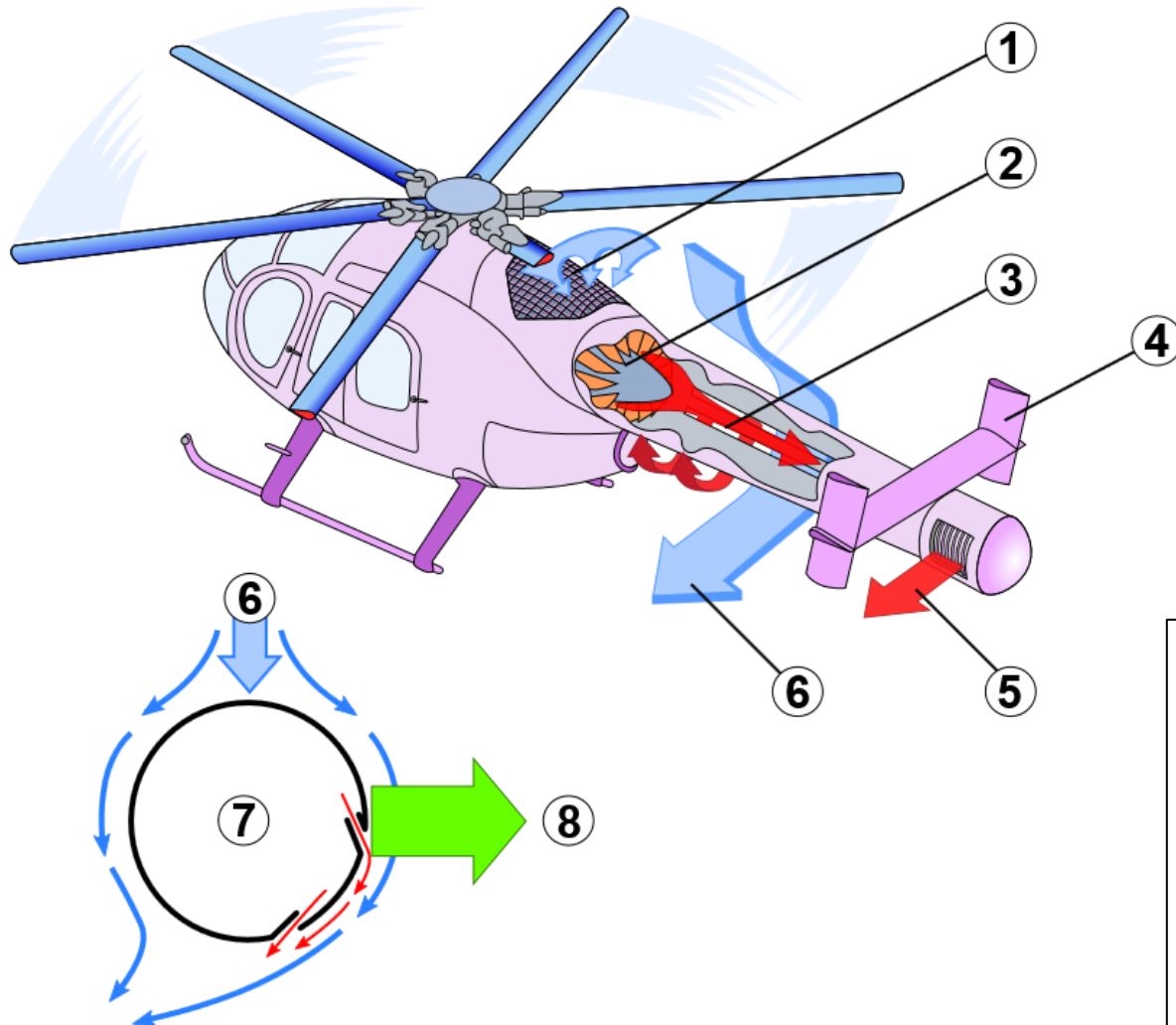
$$\begin{aligned}\dot{\theta}_y(t) &= \dot{\theta}_y(0) + \frac{1}{I_{yy}} \int_0^t T_y(\tau) d\tau \\ &= \dot{\theta}_y(0) + \frac{K}{I_{yy}} \int_0^t (\psi(\tau) - \dot{\theta}_y(\tau)) d\tau\end{aligned}$$

Note that the angular velocity appears on both sides, so this equation is not trivial to solve.

# Helicopter question

- What other effect has the tail rotor torque?
- Think about how to counteract that
  - New control strategy, new actuator on the helicopter

# NOTAR helicopter



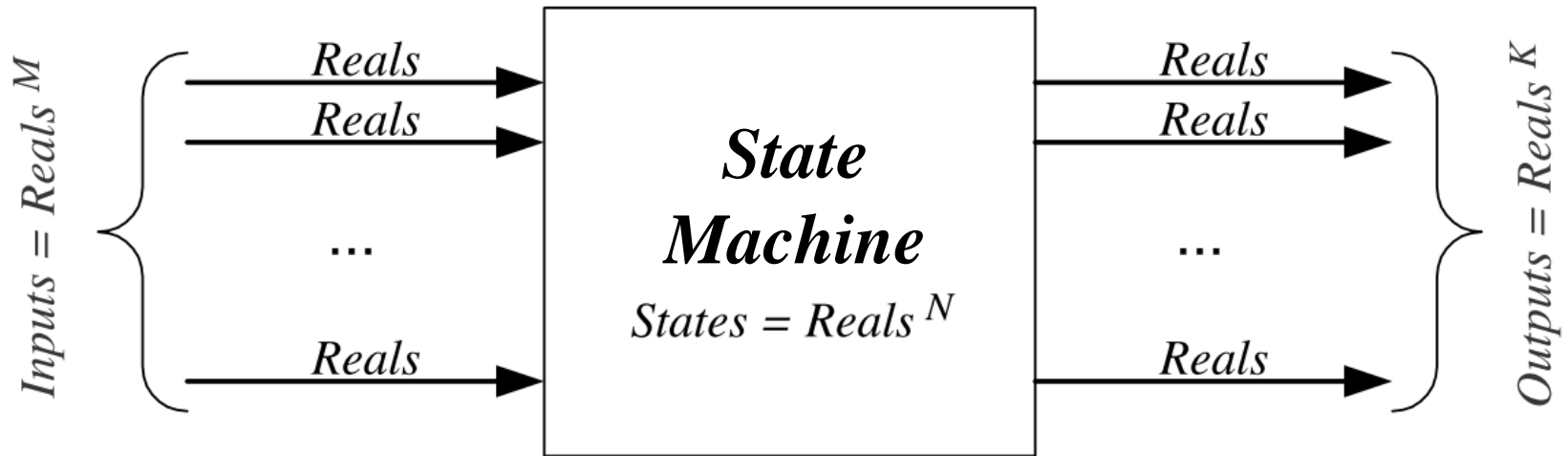
- 1 Air intake
- 2 Variable pitch fan
- 3 Tail boom with Coandă Slots
- 4 Vertical stabilizers
- 5 Direct jet thruster
- 6 Downwash
- 7 Tailboom cross-section
- 8 Anti-torque lift

# Embedded and Cyber-Physical Systems

- Modeling discrete behavior -



# Discrete model of systems



*StateMachine* = (*States*, *Inputs*, *Outputs*, *update*, *initialState*)

$s(n)$     $x(n)$     $y(n)$     $s(0)$

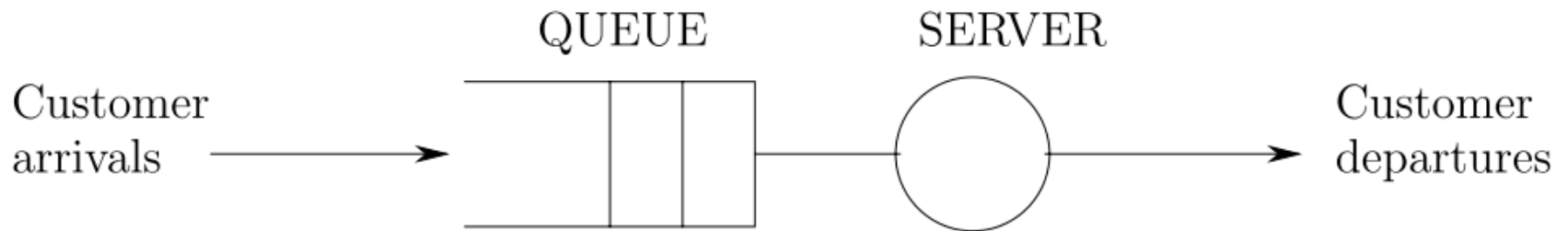
$$s(0) = \textit{initialState},$$

$$\forall n \geq 0, (s(n+1), y(n)) = \textit{update}(s(n), x(n))$$

# Discrete Event Systems

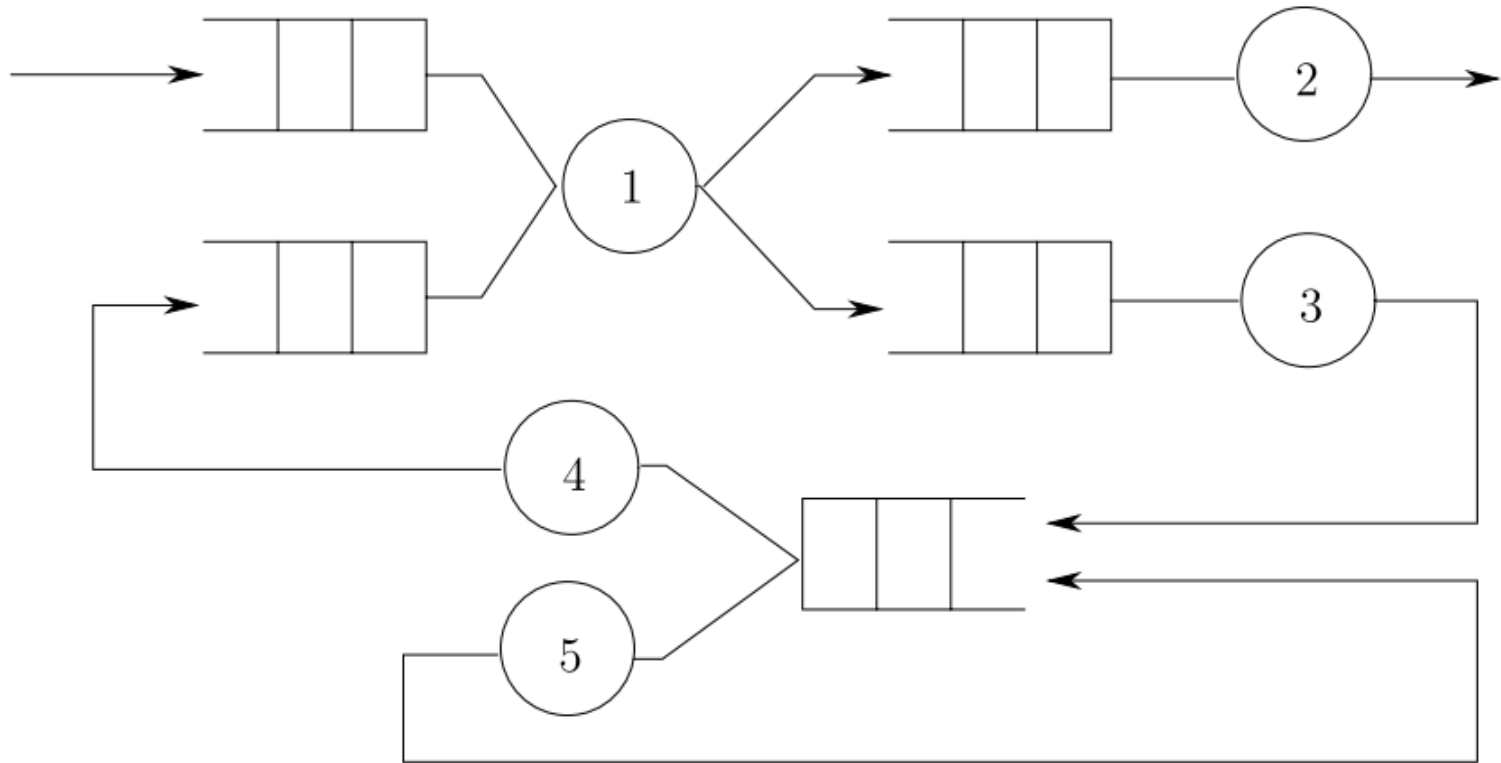
- *States* is a discrete (finite or countable) set
- *Inputs* and *outputs* are event sequences
- The *update* function is a transition between discrete states
  
- The system reacts to discrete events, not to the passage of time

# Example: Queueing systems



- Elements:
  - Capacity
  - Queueing policy
  - Arrival event
  - Departure event
- What are the states?

# Example: Queueing networks



# Other examples

- Computer systems (jobs competing for processors and peripherals)
- Manufacturing systems (production parts competing for machines)
- Traffic systems (vehicles competing for space)
- Database systems (maintaining consistency under concurrent transactions)
- *All the above are human artifacts!*

# DES as abstractions of continuous systems

- Another tank example
  - A tank with variable outflow
  - A 2-speed pump
  - Keep level between  $l_1$  and  $l_2$
- What are the states?
- What are the events?
- Do we need to represent time?

